

SUPPLEMENTAL TYPE CERTIFICATE

10078032

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

SWISS HELICOPTER ENGINEERING AG

VIA AEROPORTO 17
6596 GORDOLA
SWITZERLAND
EASA.21J.730

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.R.133

Type Certificate Holder: KAMOV JOINT STOCK COMPANY

Type: Ka-32

Model: Ka-32A11BC

Description of Design Change:

Installation of an additional bubble window on the right-hand side.

Installation of a bubble window with enhanced curvature for the right door of a Kamov KA-32A11BC, which is already equipped with a smaller radius bubble window as original installed part.

EASA Certification Basis:

The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 21 December 2021



David SOLAR

Head of Department – General Aviation & VTOL



Associated Technical Documentation:

- To be installed i.a.w. Master Document List No. 10_SHE_MDL_2019011_rev0, and Installation Bulletin No 6.2 SHE Installation Bulletin 2019011 rev0,
- To be operated i.a.w. SHE AFMS Doc. 11.1.1_SHE_AFMSUPPL_2019011_rev1,
- To be maintained i.a.w. SHE AMMS Doc. 11.1.2_SHE_AMMSUPPL_2019011_rev1

Or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -